Manual



BLACKLIGHT



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Please read the manual before you fly your new U-Turn BLACKLIGHT

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All technical details in this manual have been carefully checked by U-Turn. However we like to mention that we don't take any liability for possible mistakes, neither in legal responsibility, nor in liability cases that derive from mistakable details. We preserve the right to change this manual in any way to achieve technical improvements.

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U-Turn your airline

U-Turn GmbH was incorporated in 2002 by Thomas Vosseler and Ernst Strobl after some years of market analysis. Vosseler, hobby-pilot and successful entrepreneur in the computer and software business, is the sales and marketing specialist, while Strobl is in charge as Head of Development.

The company grew fast in Germany and Austria, and in 2004 the international distribution started. Today U-Turn gliders and related products such as rescures, helmets or flight-wear are available all over the world. The company's headquarter is in Tuningen near the Black Forest and 30 minutes by car to the lake Constance.

U-Turn paragliders are in a class of their own. U-Turn doesn't compromise on safety, and uses the best quality components and hallmark flight characteristics. Congratulation on you purchase of U-Turn glider, as it is the brand for those who appreciate the diffrence. The laws of physics are well defined. We aspire to achieve to possible within the framework of its laws. We admit this is ambitious but you will always find U-Turn at the cutting edge of technology. As Oscar Wilde once said in this very British understatement: "His taste is very basics; only always the best is good enough." The U-Turn team embodies this attitude; "We always want to deliver the best possible glider". Nothing more and most certainly nothing less. U-Turn staff takes notice of its customer wishes, so we appreciate any comments or feedback!

Please feel free to contact your competence center or U-Turn directly for any advice or direction.

Thank you

The U-Turn team would like to congratulate you on the purchase of your new U-Turn paraglider. You have made an excellent choice. We wish you long and enjoyable flights and many happy landings with your U-Turn BLACKLIGHT.

The research and Development team at U-Turn can proudly look back at many successful years in the flight sport industry. Our own concepts not only meet but exceed industry standards. The combination between the latest computer based technology and the know-how of experienced test pilots and professional competition pilots provides an excellent basis for quality. We certainly keep our customers need in mind, and always appreciate your input and constructive criticism. Should any questions occur, please don't hesitate to ask your U-Turn dealer or the U-Turn team.

In order to provide you with the latest information on technical development and inovations at U-Turn, we ask you to complete the questionnaire attached. Please mail it to the following address:

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Have fun and we wish you many flights on your new U-Turn BLACKLIGHT, your U-Turn Team



Please study this manual extensively, there is an obligation toward this recreational aircraft and its user manual to inform yourself about its specifc features prior to its first use.



We composed this handbook, in order to make the handling of your new U-Turn BLACKLIGHT as safe and easy for you as possible.

The U-Turn BLACKLIGHT

"To say it straight away: All of the certification flights are made with the serial line setup," says Strobl. The designer dislikes certification with so called folding lines in the EN-B class because they fake a safety level not existing in reality. But obvious is, that the BLACKLIGHT has a very well-thought-out line configuration. "For the main lines we use only Aramid/Kevlar lines in 11 different diameters" so Strobl. Each line is calculated exactly to its demanded strength to use the maximum potential of the wing. "The thinnest line on the wingtip got a diameter of 0.55mm, the strongest one in the middle of the wing 1.7mm." Because of the 2-lines system of the wing the total line length could be minimized to 211meters for BLACKLIGHT S and 267m for BLACKLIGHT L.

The BLACKLIGHT is- like the successful PASSION in the EN/LTF C class- equipped with the High Definition Profile (HDP). HDP means thin plastic rods on the topsail which cover from 45 to 80% of the profile depth. The optimum of this technology was found in numerous test flights in different prototype design steps.

The system corresponds perfectly with the High Pressure Crossport Design (HPCD) which means optimal designed cross ventilation openings in the ribs, which gives a dynamic handling but with a high stability. This technology is borrowed from the THRILLER and is now used in serial gliders.

The BLACKLIGHT got 50 cells and an aspect ratio of 5,8. Another fact for the high performance of the wing is the use of supreme NCV fabric material. In the topsail type 40gramm is used in water repellent quality and Skytex type 36 in the bottomsail. Elaborately is the conception of the ribs, there is type 32 Skytech material in hardfinish-quality build in.

The PPN-system instead of the heavy nose mylar gives easy starting characteristic, "and thermaling the wing is just a dream" so Strobl. The BLACKLIGHT turns willing in the thermal updraft and gains rapidly altitude. "In the end I can only advice to test the BLACKLIGHT and to experience the glide of 10 on an EN/LTF B wing."

The wing is certified in the known wide U-Turn weight range. These are 60 - 90kg for BLACKLIGHT S, 75-100kg for BLACKLIGHT SM, 85-110kg BLACKLIGHT M and 100-125kg for BLACKLIGHT L

Usage

The BLACKLIGHT is only designed for solo usage. The BLACKLIGHT is a light aircraft with a mass of less than 120 kg in the class of paragliders.

Motorised Paragliding

The U-Turn BLACKLIGHT is suitable due to it's excellent launching characteristics and uncomplicated handling in flight and high trim speed, especially good for motorized use.

Please keep in mind is that during motorized use NO ACRO MANEUVERS are allowed.

The extreme high wing loading due to the additional weight of the engine will push the U-Turn BLACKLIGHT to the brink of its load limits.

No U-Turn BLACKLIGHT has a motorized endorsement.

⚠ When flying with motor it is not allowed to exceed the weight range!

Winching

Because of its excellent starting characteristics, the U-Turn BLACKLIGHT is well suitable for winching operations. Take the following points into account:

- maximum linetension for winching is 100kp.
- if not operating at you usual winch, get acquainted with the local procedures and get a good briefing by a local pilot.
- never winch the BLACKLIGHT with loads outside the allowable weight range.
- all involved persons, machines and accessories have to have the appropriate licenses, approvals, certifications for winching.

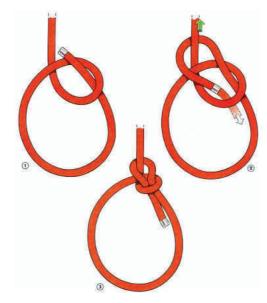
Baseline and brakeline adjustment

The factory brake-line setting corresponds to 0-free travel plus 5 cm. It is recommended to adjust your brake line travel after the first flight to your personal preferences. Be aware not to adjust the brakes too short, otherwise the glider may fly with a little, but continuous applied brake pressure. This could be extremely dangerous during takeoff, flight and landing!

The afore mentioned factory brake setting allows for ample brake travel in extreme flight situations as well as for landing. At the same time it enables during flight at trim-speed a position of comfort for the pilots arms.

In no case the setup A, B and C main lines should be changed before the wing has been flown in the original setup.

Please also note that adjusting the height of the suspension to the hangpoints on the harness, changes the relative braking travel. When setting the adjustment it is to be made certain that both sides are symmetrical and that a permanent knot is used. The bowline works particularly well because of the fact that it weakens the lines the least with excellent slip resistance.



Pfahlstich

Safety Precautions

We recommend the following precautions:

- Make your maiden flight in a familiar flying site and calm conditons.
- Test your U-Turn BLACKLIGHT only over water.
- In a "dynamic flight" are not only you exposed to high loads but also the glider. Please don't under estimate this.
- Only fly the BLACKLIGHT with at least one reserve parachute.
- observe and abide to the local aviation laws which rule in the respective country in question.
- Successful completion of appropriate training/schooling, having the needed knowledge as well as the actual flight experience are a prerequisite to operate your U-Turn BLACKLIGHT.
- The use of suitable, certified and in the respective country approved accessories (helmet, harness, reserve) is a requirement for the use of the U-Turn PASSION.
- Execute before every take off a thourough inspection of your equipment (topsail, undersail, ribs, especially the lines, carabiners, buckles, cloth speed system etc.) A flight with a tear in a glider or lines can be life threatening.
- Make sure that your flying gear is in good condition and all checks are done.
- Be aware that you as a pilot have to be in a physical and mental state to control each flight unimpaired. You have to concentrate completely on flying, in order to avoid potential distressing flight conditions. Most accidents are caused by pilot error.
- Never fly in close proximity to high voltage transmission lines, airports or motorways, over people or with lightning! You cold endanger your life and the physical well being of yourself as well as third parties and at the same time act reckless and negligent. At nor circumstance should the minimum distance fall below 50m at any give time. At airports this minimum distance to maintain is 5km..
- Inform yourself on the weather forecast and/or the predominating local weather conditions. Usethe U-Turn BLACKLIGHT only in wind strengths, in which you are able to control the wing for 100%. Do not use the U-Turn BLACKLIGHT, in wind with a great gust factor. Never use the glider with approaching thunderstorms or if probability of those of the development of thunderstorms is high. Land with thunderstorms approaching near immediately!
- The flying of aerobatics is generally forbidden and is dangerous. Unforeseen flight orientations can occur, which can spill out of control, arising the danger of overload on pilot and equipment.

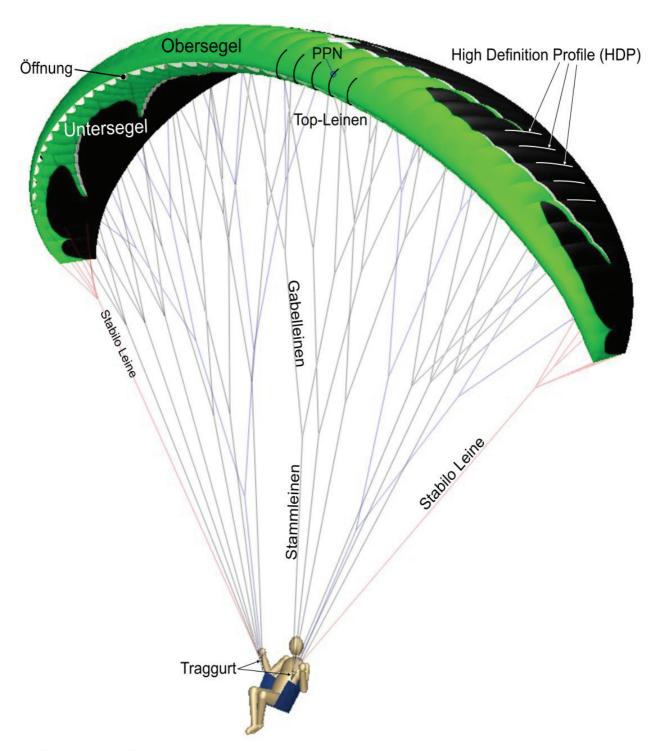
Ignoring one or several safety precautions can lead to a leisurely fun flight,

turning into a fatal event.





Glider description

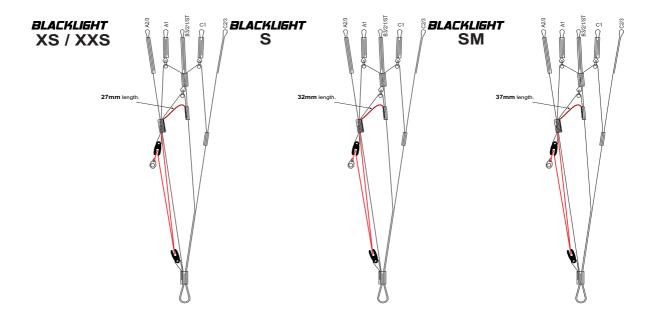


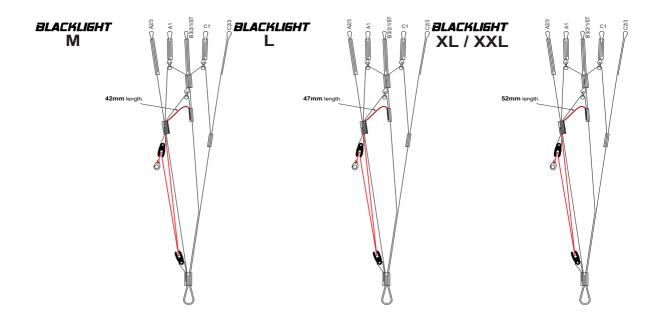
u-turn.de

Risers

The A- and B-risers have a di\$ erent colour to ensure positive identification at take off and during a B-stall descent.

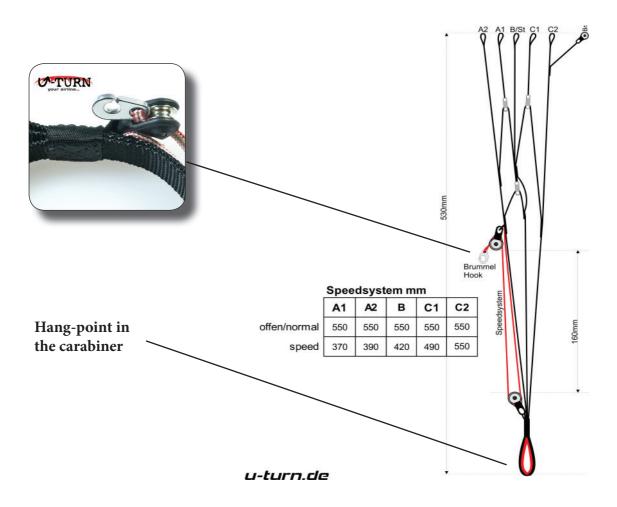
The webbing in the risers of the U-Turn BLACKLIGHT consist of sturdy stretch resistant 12mm kevlar polyester webbing, in order to warrant a long-term stable trim.





Speed System

The U-Turn BLACKLIGHT is equipped with a very effective foot actuates speed system



It increases the speed when applied with to approx. 12 km/h, depending upon wing size and pilot weight or surface loading.

Therefore it should not be activated in extreme flight situations or deactivate immediately when their occuring. All extreme flight attitudes (e.g. a-collapses) happen at accelerated speed more dynamically. Since the maximum acceleration is part of the safety behavior of the glider, it can happen that with some harnesses the speed bar to full speed cannot be used.

Operation

This instruction manual only pays attention to those points of flying technique which are important for the U-Turn BLACKLIGHT. It is not meant to substitue a basic flying education in an approved flying school! If a lying education and the appropriate experience is missing, paragliding is dangerous to life. Of the BLACKLIGHT should be flown exclusively by experienced pilots.

The Flight

Take off

After the paraglider is unpacked and laid out in the shape of a horseshoe, the following points are to be considered:

- The paraglider should be laid out in such a way that when pulling up by the A-risers, the
- center lines are evenly, and earlier tensioned than those towards the wing tips. This ensures an easy and symmectrical inflation at launch.
- Take into consideration the wind direction when laying out, so that when pulling up into the wind, both sides of the paraglider can rise symmetrically.
- Ensure the risers are without twists, and the brake lines run freely through the pulleys to
- the trailing edge of the glider.
- No lines should pass underneath the sail. A line-over at take-off can have fatal consequences.
- The 5-point check shouldn't be forgotten of course.

The center of the glider in the U-Turn BLACKLIGHT is marked by the U-Turn-logo on the leading edge. It suffices to only hold the main A-risers in the hand. Since the U-Turn BLACKLIGHT has only minor tendency to overshoot, it requires only minimal brake input during launch. If needed, directional corrections with the brakes should be undertaken only if the wing already is overhead, since too much brake input could drop the glider back. The other risers should, during take off, be left alone. With an even pull but overall light input only, the glider is to be inflated. Unlike other gliders, it is not necessary to inflate the U-Turn BLACKLIGHT with aggressive pulling or even fast running.

This is also true with little or no wind. Measured pulling up is the simplest and safest way to launch the U-Turn BLACKLIGHT. Once the pilot made sure that the glider is overhead and fully inflated, the final decision is made wether to take off . After some dynamic steps the pilot takes off.

Turning

The U-Turn BLACKLIGHT has a normal agility and reacts directly and instantly to steering inputs. You can fly flat turns with little altitude loss by shifting of bodyweight. A combination of appropriate pull on the inner brakeline and shift of bodyweight is the best way for a coordinated turn. The Turn radius depends on the amount of pull on the brakeline. At about 75% of brakeline travel, the U-Turn BLACKLIGHT increases bank significantly and performs a fast sleep turn that can be continued to a diving spiral. The diving spiral has to be initiated and terminated slowly. The bank angle is controlled by increasing and decreasing the pull on the inner brakeline.

Warning: A rapid pull on the brakeline may cause a spin.

Active Flying

The U-Turn BLACKLIGHT should be flown with light braking on both sides when there is turbulent air. An increase in angle of attack provides better stability. When entering heavy thermals or strong turbulences be mindful of that the canopy does not get behind the pilot. To avoid that, release the brakes a bit to get an increase in speed when entering the updraft.

If the canopy gets in front of the pilot when leaving a updraft or entering a downdraft the brakes have to be applied to counter that. Accelerated flight however is advisable when flying through downdraft zone. The U-Turn BLACKLIGHT is naturally very stable due to the way it's constructed. Collapsing and deforming of the canopy can be avoided by active flying (as above mentioned) in turbulent air.

Landing

Start your landing preparation at sufficient altitude. Due to its excellent flaring characteristics, the BLACKLIGHT is very easy to land. Glider in fairly normal to a straight- in final against the wind and get up in the harness early enough. According to the wind, the brakes have to be pulled firmly and dynamically, about one meter above ground, beyond the stalling point. If there is a strong headwind, be careful with the amount of braking. Don't perform landings out of steep turns and big directional changes short prior landing, to avoid PLF.

During a strong wind takeoff attempt, ground handling and landing tha leadingwdge can hit the ground with high speed. Avoid this! Otherwise the ribs, the sewings or the fabric can be damaged.



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Rapid Descent

In any situation where you have to get down ASAP for different reasons (weather, extreme updraft, or other dangers,) there are a couple of techniques that are described in this chapter.

Caution: The described manoeuvres stress you paraglider more than normal and should only be performed for practise or in a real emergency!

"Big Ears"

Pull down simultaneously both designated outer A1-risers (by grabbing at or above the maillons), about 15-20cm to fold in the wingtips. The brake toggles are to be held in hand together with the pulled down A-risers. For additional stability and for an increased sinkrate the speedsystem should be actuated. The glider remains fully steerable by weightshifting and descents at a rate of 4-7m/s straight forward.

Once you release the A-risers, the folded wingtips reinflate automatically, if not, you may pump the brakes gently. "Big earing" is due to the high wingload a very stable flight condition and well suited evine in turbulent air.

Please be aware that you reduce the trimspeed during "big ears", but this can be compensated by applying speedbar. "Big earing" in combination with weight shifting in order to get the glider to spiral dive, will achieve the highest sink rate. This descent method is often taught in SIV training. Be mindful this exposes the glider to extreme loads, should one need to use this maneuver again we recommend an equipment inspection.

B-Stall

Another very efficient descent method is the B-stall. It allows for a rate of descent of 6 to over 9 meters per second. Check the airspace under and behind you prior to initiating a B-Stall. To initiate it you hold the two B-risers above the lines carabiner. While holding the brakes in your hands at all times, pull the B-risers down progressively and symmetrically down to the shoulder to about chest level. Hold this position. Your sail will stop! ying forward, partially empty, and stabilize itself overhead. During this the wing will fall back a little, which shouldn't tempt you to release the risers again. The glider would then shoot forward and oscillate vigorously. Only once the glider has stabilized overhead it is ok to exit the B-line stall. For this release the risers symmectrically into their original position.

We recommend no to simply let the risers snap shut as this puts a lot of pressure on the material. In the paragraph titled "advanced handling" you can read what to do if you get caught unexpectedly in a stall.

Advanced Handling

Even with its high stability and good flight characteristics it is possible that the BLACK-LIGHT gets into an extreme flight condition due to pilot mistakes or turbulent air. To be prepared for such situations and able to handle them in a calm and superior manner it is best to take part in a flight safety course. Advanced manoeuvers may only be flown at sufficient altitude, in calm air and with professional supervision (i.e. during a safety course). Once again we mention that a rescuesystem is required by the law.

The following extreme manoeuvers can be either caused intentionally, by pilots mistakes or by bad weather conditions. Every pilot can get in such a situation! All mentioned extreme

manoeuvers are dangerous if they are performed without the appropriate knowledge or enough altitude or the necessary introduction. A wrong execution of these manoeuvres may have fatal consequences!

Spiral Dive

Like a normal turn, it is very easy to get the BLACKLIGHT into a spiral dive. The spiral dive gets you a descent rate up to 20 m/s. To prepare oneself in case of, practise it in optimum conditions. The diving spiral gets the pilot down faster than other techniques and is therefore best suited for an emergency descent. They move down vertically within the airmass. Don't forget the G-forces when diving down, and take that into consideration before initiating a rapid descent.

Caution: If initiation is too fast there is a danger of a spin, in this case release the brake an try a smoother initiation.

Caution: The symmetric spiral is exclusively a maneuver for amusement. It's often recommended as a descent method, however we do not recommend to do so. This is because no spiral dive can be initiated in strong thermal conditions.

Wingovers

The pilot has to perform right and left turns with increasing bank until the desired angle is reached. Collapsing wingtips is prevented by gently applying brake pressure in the up and/ or downswing of the wingover.

Full Frontal

A negative AoA caused by turbulences or the simultaneous pulldown of the A-risers by the pilot, results in a frontal collapse of the leading edge. The BLACKLIGHT comes out of a frontstall by itself very quickly. Smooth and symmetric applying of the brakes assists the opening of the canopy positively. Even, symmetrical, subtle pumping of the brakes can assist the reopening.

Collapses

Even with its high stability and very responds well in turbulence, strong turbulences can cause the BLACKLIGHT to collapse. That situation is not really dangerous and clears itself automatically, without any further input required. To support the recovery, firmly apply brakes on the affected side and simultaneously steer opposite on the open side. When a large part of the canopy is collapsed be careful and smooth when applying opposite steering to avoid a complete disruption of airflow and entering a fullstall.

In case of larger deflations the counter steering is to be excercised with restraint / in moderation, in order not to completely interupt the airflow to the positive side of the wing and spin the glider.

How to avoid collapses

Single side collapses close to the ground are the number one reason for accidents with paragliders. To avoid them, or how to handle the situation when it happened, some tips and tricks from U-Turn test- and competition pilot Ernst Strobl:

The best way to avoid collapses upfront is the right choice of the paraglider. A lot of pilots fly a glider that is a little too hot to handle for them. So why don't you get a glider with a lower rating but in the end fly better and higher in the updrafts and have a lot more fun and by the way be safer, too. To optimize the feeling for your glider on the ground, try the following:

Pratice on the ground with the right wind at a suitable location. Slowly pull up the canopy and try to hold it up as long as possible without looking towords?? it. That is a good way to improve the feeling for your glider and is a prerequisite for "active flying" (the key to avoid collapses). Very important is also a close look at the terrain. Watch for obstacles that could cause turbulences (buildings, trees, ...). On certain days, for example a freshly mowed madow as landing field, could cause a lot of thermal activity. Fly very alert on a thermal active day. Watch your canopy, collapses most of the time, announce themself. Light braking in turbulences mostly avoids a collapse. You should have already pratised that on the ground. Should a collapse occur close to the ground don't always try to prevent a turn away. There is a danger when the braking on the open side is to strong, to lose the airflow on this side and stall the glider. Rather use the turn away motion to try to open the collapsed side. Apply smooth braking on the open side, depending on the size of the collapse, and maybe a little pumping action. Some canopies open a lot better when the brakes are fully applied once on the according side, but that depends on the brakelines adjustment and your armlength. Wrapped lines are cleared by braking the opposite side at enough altitude and pumping the affected side a couple of times. Watch out for a possible stall. If that does no clear the situation, try to pull dow the outer lines as much as possible. If you are too low for that, stabilize the canopy on the opposite side avoid turning away, and leave the lines like they are. Instead of any - risky manoeuvers rather concentrate on the landing. In the end one more advice in order to have all kinds of situations under control.

Visit a safety-training above water. There is no better way to practice the right behaviour than simulating a dangerous situation. Don't get caught off guard by your first collapse. In addition, during safety-training you can familiarize yourself with the particulars of your equipment and you gain confidence in your gliders as well as your own abilities.

Thus far the expert advise concerning collapses, by Ernst Strobl.



Deep Stall

The U-Turn BLACKLIGHT is not stall sensitive. If in a stall, caused by overpulling on the brakes, the rear risers or a delayed B-stall exit, the release of the brakes or the rear risers, recovers the stall. Should the stall be caused by an extreme flight condition or configuration (i.e. takeoff weight to low), a symmetric forward push on the A-riser or step the speed system recovers the stall.

Warning: Practicing stalls should be done with enough safe altitude. Never apply asymmetric brakes during a stall, it could cause a spin. If the BLACKLIGHT in deep stall, one should only brake release the brake the glider is in front.

Fullstall

To initiate a full stall, pull both brakes without a wrap slowly to the point of stall. As soon as the point of stall is reached, hold both hands down. The glider falls back. At point, under no circumstance should the hands let up or release the brakes. To recover from a full stalls the canopy should be stabilized overhead and prefilled. For this slightly let up both brakes symmetrically. To exit completely, let up both brakes symmetrically and slowly in its entirety. With a correct symmetrical exit the glider returns swiftly, as soon as the glider shoots strongly forward, it must be checked by a brief brake input. An asymmetrical recovery is to be avoided, this could lead to falling into the glider.

Negativ Turn

A negative turn/spin is initiated, when the pilot pulls the brake on one side fast and completely though the point of stall while letting the other side of the wing fly freely. With a negative turn the glider turns relatively fast around its center, while the inside filies backwards. In order to exist a spin, the applied brake released, where stalled side of the wing can pick up speed or one exit though a full stall, by braking the flying side into a stall also.

Warning: The Spin and the Fullstall and unpredictable and dangerous flight attitudes and should only be executed in a safety training under guidance and never be intentionally executed. There is danger of riser twist. With a riser twist the brake lines can get blocked.

Warning: The glider has been overloaded. Fullstalls and negative turns / spins as a descent method is dangerous, because a wrong exit, independent of glider type, can have fatal consequences.

Emergency Piloting

In any situation where normal steering with the brakelines is not possible, the U-Turn BLACKLIGHT can be steered with the back risers easily. Turns can be flown with weightshift, however be mindful that the glider doesn't lock into a spiral.

Transport and storage

When transporting the glider dont expose it to any liquids and has to be packed totally dry. Store the BLACKLIGHT always protected from UV radiation. Furthermore never store the wing together with acids or similar gods. A dry storage is very important!



Maintenance and Care

Because U-Turn only uses high quality materials, your BLACKLIGHT will be airworthy for many years if you take good care. The aging of your BLACKLIGHT depends on the total flying time, the conditions you fly in, the amount of UV radiation it is exposed to and the intensity and quality of care. A couple of tips for maintenance and care:

Long lasting exposure to UV radiation and normal use stress the material

- Don't expose your glider to the sun when there is no need to
- Consider the choice of terrain where you lay out the glider for takeoff
- Assymmetrical and changing folding patterns prolong the lifetime of the material especially in the middle section.

Please take following points into consideration

- regular checks for damage
- no unneccessary bending
- lines after overloads (tree landing, water landing, etc.) for its strength and correct length to be checked and exchanged if necessary
- in case of changing inflight handling characteristics, the line have to be checked for their-correct length
- don't tie the brakelines on the grips if not needed, it weakens the lines

To clean the canopy use warm water and a soft sponge. If you use a detergent for hard stains, make sure that you rinse intensively afterwards. Never apply any chemicals for cleaning, they weaken the material and damage the coating. Store your glider at a dry and dark location away from any chemicals. After two years or 300 flighthours, whichever occurs first, your BLACK-LIGHT has to be inspected by the manufacturer, in case of extreme use we are glad to do that earlier. Only you know about the condition of your glider. Should there be a need for any repairs they are to be done by the manufacturer.

Nature and environment friendly behaviour

We ask you to perform our sport in a manner, that impacts nature and environment with minimum intensity. Please do not walk beside marked paths, don't leave any waste, please be not noisy and respect the sensitive biological equilibrium in the mountains. Especially at starting areas maximum care for nature is necessary. The synthetic materials your U-Turn glider is build must be depolluted appropriately. Please send your U-Turn glider at the end of its lifecycle back to U-Turn. We will take care for recycling and removal.

FLight accessories

Harness

All officially approved harness systems with mounting about the breast height are suitable for the BLACKLIGHT. The lower the mounting, the better is the steering by shifting of the bodyweight. U-Turn recommends the new IQ4 harness for its highest level of safety and convenience. The positioning of the mounting also changes the relative brakedistance. If you have any questions about the usage of your harness with the BLACKLIGHT, ask your U-Turn dealer or directly contact U-Turn. We assist you in any possible way.



Suitable Rescue System

It is required by law and absolutely neccessary for safe operation of your paraglider that you always carry a rescue system with you. When choosing a rescue system, watch out that it is approved and suitable for the intended takeoff weight. With the innovative rescue systems of the SECURE-series by U-Turn light-weight, convenient and safe reserves are available. The SECURE rescues offer extremely short opening times and low sink-rates.







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Assumption of Risk

Flying the U-Turn BLACKLIGHT is inherent with certain dangers of bodily harm or even death of the user of this product or third party equipment. With the use of the BLACKLIGHT you assume all known and unknown risks and accept probable and improbable risks to injury. The dangers innate with the practice this kind of sport can be reduced by adhering to the warning notes in the manual, as well as the required attention to detail on each flight. The risks inherent to the sport can be reduced to a large degree, if one adheres both to the maintenance guidelines, which are listed in this operating manual, as well as using common sense.

Liability claim and renouncement of exclusion

With the completion of the sale of a U-Turn BLACKLIGHT you express your in consent with the following points of legal specifications:

THE RENOUNCEMENT EXCLUSION OF ALL LIABILITY CLAIMS

deriving from the use the U-Turn BLACKLIGHT and or either compenents thereof, now or in the future, against the U-Turn GmbH and all other contracting parties, that could arise.

Releasing U-Turn GmbH and all other contracting parties of all liability claims concerning loss, damage, injury or expenses to you, your next of kin, relatives or any other user of the U-Turn BLACKLIGHT as a result could suffer. This includes but is not limited to lawful or contractual liability on behalf U-Turn GmbH and all other contracting parties as a result of the of production and processing the U-Turn BLACKLIGHT and all its components. With the occurrence of death or disability, all directives stated here come into force and bind their beneficiaries, next of kin, trustees, legal successors and/ or representatives. The U-Turn GmbH and all other contracting parties express no verbal or written representation and denial expressively that this was done, with exception of what is speci% ed in and in the manual the U-Turn BLACKLIGHT.

Safety Advices and Liability

This glider complies with EAPR, AFNOR (SHV and ACPUL) regulations, for the tested type, at time of delivery (see appendix). The operation of the glider is at your own risk. The manufacturer and the dealer don't take any liability for accidents and follow on damages. Please consider all safety notes, cautions and warnings for safe flying. Further, we assume that the pilot has the necessary certifications and that the legal limitations are being followed. Use of the equipment is at your own risk. Follow the safety instructions for a safe flight.

Release of Liability, Renouncement of Entitlement

Hereby you declare, that you -prior to use of the U-Turn BLACKLIGHT- the U-Turn BLACK-LIGHT user manual in its entirety, including directions and warnings, which are included in this user manual, have read and understood. Moreover to carry responsibility - prior to granting the use by a third party of U-Turn BLACKLIGHT - through transferring ownership temporary or permanently, for this other user to have read and unterstood the U-Turn BLACKLIGHT user manual in its entirety, including directions and warnings, which are included in this user manual.

| Date | Signature first Pilot |
|----------|------------------------|
| Date | Signature second Pilot |
| Date | Signature third Pilot |

U-Turn cannot be hold responsible for any 2-year inspection and any repairs not performed by U-Turn or an U-Turn authorized dealer.

Any checking or repairing performed by people not authorized by U-Turn will cause denial of any warranty!

Technical Data U-Turn BLACKLIGHT



| | | | | Technical Data | | | | |
|--|---|---|---|---|--|---|--|---|
| | XXS | XS | S | SM | Σ | _ | χĽ | XXL |
| Start Weight | 40-80 kg | 50-80 kg | 60-90 kg | 75-100 kg | 85-110 kg | 100-130 kg | 115-145 kg | 130-160 kg |
| Flat Area | 19,5 m² | 21,5 m² | 23,5 m² | 25,5 m² | 27,5 m² | 29,5 m² | 32,0 m² | 35,0 m² |
| Projected Area | 16,47 m² | 18,16 m² | 19,85 m² | 20,54 m² | 23,23 m² | 24,92 m² | 27,03 m² | 29,56 m² |
| Flat Wingspan | 10,63 m | 11,16 m | 11,67 m | 12,16 m | 12,63 m | 13,08 m | 13,62 m | 14,24 m |
| Projected Wingspan | 8,25 m | 8,66 m | 90'6 m | 9,43 m | 9,8 m | 10,15 m | 10,57 m | 11,05 m |
| Flat AR | 5,8 | 5,8 | 5,8 | 5,8 | 5,8 | 5,8 | 5,8 | 5,8 |
| Projected AR | 4,13 | 4,13 | 4,13 | 4,13 | 4,13 | 4,13 | 4,13 | 4,13 |
| Chord: Center / Wingtip | 2,265 / 0,565 m | 2,378 / 0,590 m | 2,486 / 0,620 m | 2,590 / 0,646 m | 2,689 / 0,670 m | 2,785 / 0,695 m | 2,901 / 0,723 m | 3,034 / 0,757 m |
| V-Trim | 39-40 km/h | 39-40 km/h | 39-40 km/h | 39-40 km/h | 39-40 km/h | 39-40 km/h | 39-40 km/h | 39-40 km/h |
| V-Max | 51-52 km/h | 51-52 km/h | 51-52 km/h | 51-52 km/h | 51-52 km/h | 51-52 km/h | 51-52 km/h | 51-52 km/h |
| Bridle height | 6,381 m | 6,706 m | 7,005 m | 7,297 m | 7,578 m | 7,848 m | 8,174 m | 8,301 m |
| Nr. of Cells | 20 | 20 | 20 | 20 | 20 | 20 | 20 | 20 |
| Glider Weight | 4,0 kg | 4,2 kg | 4,4 kg | 4,8 kg | 5,2 kg | 5,6 kg | 6,2 kg * | 6,7 kg* |
| Bridle length | 208,8 m | 219,6 | 229,9 | 239,7 | 249,2 m | 258,3 m | 269,3 m | ™ 0.282 |
| Line Diameter | 0,55/0,65/0,7/1,1 1,2/1,3/1,45/1,6mm | 0,55/0,65/0,7/1,1 1,2/1,3/1,45/1,6mm | 0.55/0.65/0.7/1,1 0.55/0.65/0.7/1,1 0.55/0.65/0.7/1,1 .2/1,3/1,45/1,6mm 1.2/1,3/1,45/1,6mm 1.2/1,3/1,45/1,6mm | | 0,55/0,65/0,7/1,1 0,55/0,65/0,7/1,1 0,55/0,65/0,7/1,1 1,2/1,3/1,45/1,6mm 1,2/1,3/1,45/1,6mm | 0,55/0,65/0,7/1,1 1,2/1,3/1,45/1,6mm | 0,55/0,65/0,7/1,1 1,2/1,3/1,45/1,6mm 1,2/1,3/1,45/1,6mm | 0,55/0,65/0,7/1,1 1,2/1,3/1,45/1,6mm |
| Speed System / Trimmer | Yes / No | Yes / No | Yes / No | Yes / No | Yes / No | Yes / No | Yes / No | Yes / No |
| Certification | No Certification | No Certification | EN-B / LTF-B | EN-B/ LTF-B | EN-B / LTF-B | EN-B / LTF-B | No Certification | No Certification |
| On this indicate and processing | LTF 91/09 | LTF 91/09 | LTF 91/09 | LTF 91/09 | LTF 91/09 | LTF 91/09 | LTF 91/09 | LTF 91/09 |
| cei mieu stailual as ailu pi oceuul es | EN 926/1 & 926/2 | EN 926/1 & 926/2 | EN 926/1 & 926/2 | EN 926/1 & 926/2 | EN 926/1 & 926/2 | EN 926/1 & 926/2 | EN 926/1 & 926/2 | EN 926/1 & 926/2 |
| Folding lines used for certification | No | No | No | No | No | No | No | No |
| Certification No. | | | EAPR-GS-7575/12 | EAPR-GS-7574/12 | EAPR-GS-7538/12 | EAPR-GS-7564/12 | | |
| | | Emors and omissions expec | Errors and omissions expected. Subject to change without notice. Reproduction in whole or in part without written permission of UTurn GnibH is prohibited | production in whole or in part without writ | en permission of U-Tum GmbH is prohibit | Jad. | | |

Table of area loading and Color-Info

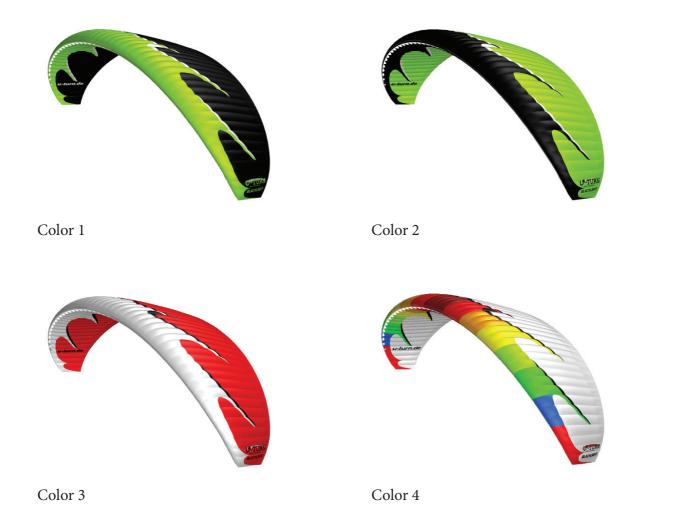


Surface load per square meter



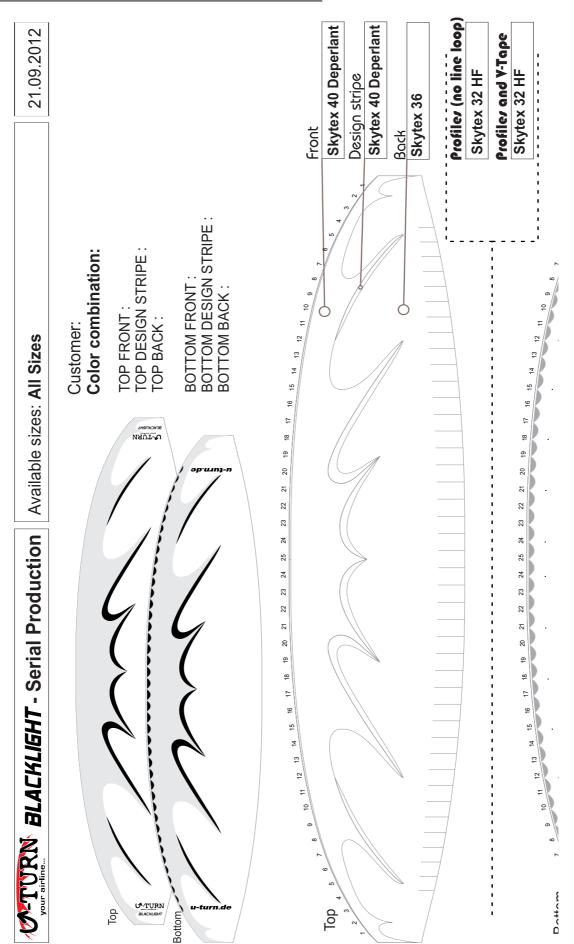
| Start Weight (kg) | 40 | 45 | 50 | 55 | 60 | 65 | 70 | 75 | 80 | 85 | 90 | 95 | 100 | 105 | 110 | 115 | 120 | 125 | 130 | 135 | 140 | 145 | 150 | 155 | 160 |
|-------------------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|------|
| BLACKLIGHT XXS | 2,05 | 2,31 | 2,56 | 2,82 | 3,08 | 3,33 | 3,59 | | | | | | | | | | | | | | | | | | |
| BLACKLIGHT XS | | | 2,33 | 2,56 | 2,79 | 3,02 | 3,26 | 3,49 | 3,72 | | | | | | | | | | | | | | | | |
| BLACKLIGHT S | | | | | 2,55 | 2,77 | 2,98 | 3,19 | 3,40 | 3,62 | 3,83 | | | | | | | | | | | | | | |
| BLACKLIGHT SM | | | | | | | | 2,94 | 3,14 | 3,33 | 3,53 | 3,73 | 3,92 | | | | | | | | | | | | |
| BLACKLIGHT M | | | | | | | | | | 3,09 | 3,27 | 3,45 | 3,64 | 3,82 | 4,00 | | | | | | | | | | |
| BLACKLIGHT L | | | | | | | | | | | | | 3,39 | 3,56 | 3,73 | 3,90 | 4,07 | 4,24 | 4,41 | | | | | | |
| BLACKLIGHT XL | | | | | | | | | | | | | | | | 3,59 | 3,75 | 3,91 | 4,06 | 4,22 | 4,38 | 4,53 | | | |
| BLACKLIGHT XXL | | | | | | | | | | | | | | | | | | | 3,71 | 3,86 | 4,00 | 4,14 | 4,29 | 4,43 | 4,57 |

Dark green: This area is the ideal stress loading per unit area.



U-Turn offers special colors too. If you have questions about your BLACKLIGHT in special colors don't hesitate to contact your competence center or directly us. We are pleased to help.

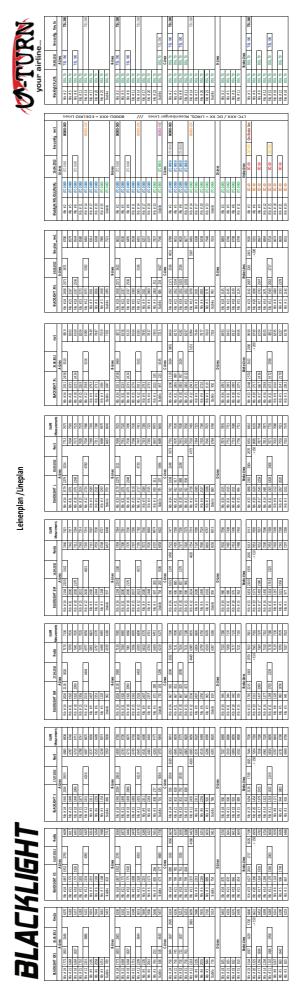
Material List U-Turn BLACKLIGHT



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Lineplan



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Certification U-Turn BLACKLIGHT S





EAPR e.V - Marktstr. 11 - D-87730 Bad Grönenbach - Germany

| | Minimum take off | weight | Maximum take off w | /eight |
|-------------------------|------------------|--------|------------------------|--------|
| Date of testing | 06.06.12 | | 16+18.06.2012 | |
| Testpilot | Mike Küng | | Hannes Tschofen | |
| Harness | EAPR-Equipment | | Academy Test Equipment | |
| Pilot's take off weight | 65 kç | | 90 kg | |

| Classification | В |
|----------------|---|
|----------------|---|



| Test-criteria | | 41066 | Evaluation | 16+18.06.2012 | Evaluation |
|--|-------------|----------------------------------|------------|----------------------------------|------------|
| 1. Inflation / take-off - 4.1.1 | | | | | |
| Rising behavior | | Smooth, easy and constant rising | А | Smooth, easy and constant rising | А |
| Special take off technique required | | No | Α | No | Α |
| 2. Landing - 4.1.2 | | | | | , |
| Special landing technique required | | No | А | No | А |
| 3. Speeds in straight flight - 4.1.3 | | | | | · |
| Trim speed more than 30km/h | | Yes | Α | Yes | Α |
| Speed range using the controls larger than 10km/ | h | Yes | А | Yes | А |
| Minimum speed | | Less than 25 km/h | А | Less than 25 km/h | А |
| 4. Control movement - 4.1.4 | | | | | |
| Max. weight in flight up to 80kg | | Increasing > 55cm | А | | - |
| Max. weight in flight 80 to 100kg | | | - | Increasing > 60cm | А |
| Max. weight in flight greater than 100kg | | | - | | - |
| 5. Pitch stability exiting accelerated flight - 4.1 | .5 | | | | |
| Dive forward angle on exit | | Dive forward less than 30° | A | Dive forward less than 30° | А |
| Collapse occurs | | No | Α | No | Α |
| 6. Pitch stability operating controls during account | elerated fl | ight - 4.1.6 | | | |
| Collapse occurs | | No | Α | No | Α |
| 7. Roll stability and damping - 4.1.7 | | | | | |
| Oscillations | | Reducing | Α | Reducing | Α |
| 8. Stability in gentle spirals - 4.1.8 | | | | | |
| Tendency to return to straight flight | | Spontaneous exit | Α | Spontaneous exit | Α |
| 9. Behaviour in a steeply banked turn - 4.1.9 | | | | | |
| Sink rate after two turns | | More than 14m/s | В | More than 14m/s | В |
| 10. Symmetric front collapse - 4.1.10 | | | | | |
| Entry | | Rocking back less than 45° | Α | Rocking back less than 45° | Α |
| Recovery | trim speed | Spontaneous in less than 3 sec | А | Spontaneous in less than 3 sec | А |
| Dive forward angle on exit | .E | 0° - 30° Keeping course | А | 30° - 60° Keeping course | В |
| Cascade occurs | ± | No | А | No | А |
| Entry | D. | Rocking back less than 45° | Α | Rocking back less than 45° | Α |
| Recovery | accelerated | Spontaneous in less than 3 sec | А | Spontaneous in less than 3 sec | А |
| Dive forward angle on exit | CCe | 30° - 60° Keeping course | В | 30° - 60° Keeping course | В |
| Cascade occurs | 70 | No | А | No | А |
| 11. Exiting deep stall (parachutal stall) - 4.1.11 | | | | | |

Flight Test-Report Stand - 08.04.2010 Seite 1

Certification U-Turn BLACKLIGHT S

| Doop stall achieved | | I Voc | | | | l Voc | | | |
|--|----------------------------------|-------------------------------|--------------------|-------------------|--------|--------------------------|----------------------|-------------------|---|
| Deep stall achieved | | Yes | | | | Yes | | | |
| Recovery | | Spontaneous in less | than 3 sec | | Α | Spontaneous in | less than 3 sec | | Α |
| Dive forward angle on exit | | 30° - 60° | 45 450 | | В | 30° - 60° | - I th 45° | | В |
| Change of course Cascade occurs | | Changing course les | ss than 45 | | A A | Changing course No | e less than 45 | | A |
| 12. High angle of attack recovery - 4.1.12 | | | | | 7. | | | | |
| Recovery | | Spontaneous in less | than 3 sec | | А | Spontaneous in | less than 3 sec | | А |
| Cascade occurs | | No No | | | A | No | | | A |
| 13. Recovery from a developed full stall - 4.1.1 | 3 | 140 | | | A | 140 | | | A |
| Dive forward angle on exit | | 30° - 60° | | | В | 30° - 60° | | | В |
| Collapse | | No collapse | | | Α | No collapse | | | Α |
| Cascade occurs (other than collapse) Rocking backward | | No Less than 45° | | | A A | No Less than 45° | | | A |
| Line tension | | Most lines tight | | | A | Most lines tight | | | A |
| 14. Asymmetric collapse (trim speed) - 4.1.14 | | | | | | | | | |
| Change of course until re-inflation | Se | < 90° | Dive or roll angle | 15° - 45° | Α | < 90° | Dive or roll angle | 15° - 45° | Α |
| Re-inflation behavior | trim speed, max 50% collapse | Spontaneous re-infla | ation | | А | Spontaneous re- | -inflation | | А |
| Total change of course | ds w | Less than 360° | | | А | Less than 360° | | | А |
| Collapse on the opposite side occurs Twist occurs | ax & | No No | | | A | No No | | | A |
| Cascade occurs | | No | | | A A | No | | | A |
| Change of course until re-inflation | 4. | | Dive or roll angle | 15° - 45° | В | 90° - 180° | Dive or roll angle | 15° - 45° | В |
| Re-inflation behavior | trim speed, max 75% collapse | Spontaneous re-infla | | | A | Spontaneous re- | | | A |
| | trim speed | · | u | | | · | | | |
| Total change of course Collapse on the opposite side occurs | trim: | Less than 360° No | | | A A | Less than 360° No | | | A |
| Twist occurs | may _ | No | | | A | No | | | A |
| Cascade occurs | | No | | | А | No | | | Α |
| Change of course until re-inflation | ě | < 90° | Dive or roll angle | 15° - 45° | А | < 90° | Dive or roll angle | 15° - 45° | Α |
| Re-inflation behavior | accelerated, max 50% collapse | Spontaneous re-infla | ation | | Α | Spontaneous re- | -inflation | | А |
| Total change of course | eler %0 | Less than 360° | | | Α | Less than 360° | | | А |
| Collapse on the opposite side occurs | ax 21 | No | | | Α | No | | | Α |
| Twist occurs Cascade occurs | Ë | No No | | | A A | No No | | | A |
| Change of course until re-inflation | a) | | Dive or roll angle | 15° - 45° | В | 90° - 180° | Dive or roll angle | 15° - 45° | В |
| Re-inflation behavior | accelerated, max 75% collapse | Spontaneous re-infla | ation | | A | Spontaneous re- | inflation | | A |
| Total change of course | elera % co | Less than 360° | | | A | Less than 360° | | | A |
| Collapse on the opposite side occurs | accelerated, x 75% collap | No | | | A | No | | | A |
| Twist occurs | ma | No | | | Α | No | | | Α |
| Cascade occurs 15. Directional control with a maintained asym | metric col | No lanse - 4 1 15 | | | Α | No | | | Α |
| Able to keep course straight | | Yes | | | А | Yes | | | А |
| 180° turn away from the collapsed side possible in | 10 sec | Yes | | | А | Yes | | | А |
| Amount of control range between turn and stall or | | More than 50% of the | e symmetric c | ontrol travel | A | More than 50% o | of the symmetric c | ontrol travel | A |
| 16. Trim speed spin tendency - 4.1.16 | | | | | | | | | |
| Spin occurs | | No | | | А | No | | | А |
| 17. Low speed spin tendency - 4.1.17 | | | | | | | | | |
| Spin occurs 18. Recovery from a developed spin - 4.1.18 | | No | | | Α | No | | | А |
| | | Stone eninning in I- | ee than 00° | | ٨ | Stone enimpine | n lees than 00° | | ^ |
| Spin rotation angle after release | | Stops spinning in les | 55 HIAN 90° | | A | Stops spinning in | n iess (nan 90° | | A |
| Cascade occurs 19. B-line-stall - 4.1.19 | | No | | | Α | No | | | А |
| Change of course before release | | Changing course les | ss than 45° | | А | Changing course | e less than 45° | | А |
| Behaviour before release | | Remains stable with | | | A | | with straight span | | A |
| Recovery | | Spontaneous in less | than 3 sec | | А | Spontaneous in | less than 3 sec | | А |
| Dive forward angle on exit | | 30° - 60° | | | A | 0° - 30° | | | A |
| Cascade occurs 20. Big ears - 4.1.20 | | No | | | Α | No | | | Α |
| Entry procedure | | Special device requir | red | | Α | Special device re | equired | | Α |
| Behaviour during big ears | | Stable flight | | | А | Stable flight | | | Α |
| Recovery | · <u></u> | Recovery through pil 3 sec | lot action in le | ss than a further | В | Recovery throug 3 sec | h pilot action in le | ss than a further | В |
| Dive forward angle on exit | | 0° - 30° | | | A | 0° bis 30° | | | Α |
| 21. Big Ears in accelerated flight - 4.1.21 | | | | | | | | | |
| Entry procedure | | Special device requir | red | | А | Special device re | equired | | А |
| Behaviour during big ears | | Stable flight | | | A | Stable flight | | | Α |
| Recovery | | Recovery through pil | lot action in le | ss than a further | В | Spontaneous in | 3 to 5 sec | | А |
| Dive forward angle on exit | | 3 sec 0° - 30° | | | A | 0° bis 30° | | | A |
| Behaviour immediately after releasing the accelar | ator while | Stable flight | | | A | Stable flight | | | A |
| maintaining big ears | | Stable Hight | | | - 7 | Casio ingit | | | |
| 22. Behaviour exiting a steep spiral - 4.1.22 | | | | | | | | | |

Certificarion U-Turn BLACKLIGHT S

| Tendency to return to straight flight | Spontaneous exit | Α | Spontaneous exit | Α |
|---|--------------------------------------|------------|---|----------------------------|
| Turn angle to recover normal flight | Less than 720°, spontaneous recovery | Α | Less than 720°, spontaneous recovery | Α |
| 23. Alternative means of directional control - 4.1.23 | | | | |
| 180° turn achievable in 20 sec | Yes | Α | Yes | Α |
| Stall or spin occurs | No | Α | No | Α |
| 24. Any other flight procedure and/or configuration des | cribed in the user's manual - 4.1.24 | | | |
| Procedure works as descibed | | NA | | NA |
| Procedure suitable for novice pilots | | NA | | NA |
| Cascade occurs | | NA | | NA |
| 25. Remarks of testpilot: | | | | |
| | | | * Beim Tragegurt des Blacklight S, muß im Gegensetz zu den anderen Größen, zwischen den A und B Tragegurten | eine fixe Verbindung sein. |
| | | | | |
| | | | | |
| | | | | |
| | | | | |
| Copyright Ralf Antz 2010 | This Flight Te | est Report | was generated automatically and is valid withou | ıt signature |

Certification U-Turn BLACKLIGHT S/M



EAPR e.V - Marktstr. 11 - D-87730 Bad Grönenbach - Germany

| | Minimum take off w | eight | Maximum take off w | veight |
|-------------------------|--------------------|-------|------------------------|--------|
| Date of testing | 28.03.12 | | 25.03.12 | |
| Testpilot | Mike Küng | | Tschofen Johannes | |
| Harness | Academy-Equipment | | Academy Test Equipment | |
| Pilot's take off weight | 75 kg | | 100 kg | |

| Classification | В |
|----------------|---|
|----------------|---|



| Test-criteria | | 40996 | Evaluation | 40993 | Evaluation |
|--|-------------|----------------------------------|------------|----------------------------------|------------|
| 1. Inflation / take-off - 4.1.1 | | | | | |
| Rising behavior | | Smooth, easy and constant rising | А | Smooth, easy and constant rising | А |
| Special take off technique required | | No | А | No | А |
| 2. Landing - 4.1.2 | | | | | |
| Special landing technique required | | No | А | No | А |
| 3. Speeds in straight flight - 4.1.3 | | | | | |
| Trim speed more than 30km/h | | Yes | Α | Yes | А |
| Speed range using the controls larger than 10km/h | n | Yes | А | Yes | А |
| Minimum speed | | Less than 25 km/h | Α | Less than 25 km/h | Α |
| 4. Control movement - 4.1.4 | | | | | |
| Max. weight in flight up to 80kg | | Increasing > 55cm | А | | - |
| Max. weight in flight 80 to 100kg | | | - | Increasing > 60cm | А |
| Max. weight in flight greater than 100kg | | | - | | - |
| 5. Pitch stability exiting accelerated flight - 4.1. | .5 | | <u>'</u> | | • |
| Dive forward angle on exit | | Dive forward less than 30° | А | Dive forward less than 30° | А |
| Collapse occurs | | No | А | No | A |
| 6. Pitch stability operating controls during acce | elerated fl | ight - 4.1.6 | | | |
| Collapse occurs | | No | Α | No | Α |
| 7. Roll stability and damping - 4.1.7 | | | | | |
| Oscillations | | Reducing | А | Reducing | Α |
| 8. Stability in gentle spirals - 4.1.8 | | | | | |
| Tendency to return to straight flight | | Spontaneous exit | Α | Spontaneous exit | А |
| 9. Behaviour in a steeply banked turn - 4.1.9 | | | | | |
| Sink rate after two turns | | More than 14m/s | В | More than 14m/s | В |
| 10. Symmetric front collapse - 4.1.10 | | | | | |
| Entry | _ | Rocking back less than 45° | А | Rocking back less than 45° | А |
| Recovery | trim speed | Spontaneous in less than 3 sec | А | Spontaneous in less than 3 sec | А |
| Dive forward angle on exit | Ë | 0°- 30° Keeping course | А | 0°- 30° Keeping course | А |
| Cascade occurs | + | No | А | No | А |
| Entry | p | Rocking back less than 45° | Α | Rocking back less than 45° | Α |
| Recovery | accelerated | Spontaneous in less than 3 sec | А | Spontaneous in less than 3 sec | Α |
| Dive forward angle on exit | эссе | 30°-60° Keeping course | В | 30° - 60° Keeping course | В |
| Cascade occurs | to . | No | Α | No | Α |
| 11. Exiting deep stall (parachutal stall) - 4.1.11 | | | | | |

Certification U-Turn BLACKLIGHT S/M

| Deep stall achieved | | Yes | | 1 | | Yes | | | | | |
|--|----------------------------------|--|--------------------------------|-------------------|-------------------------------------|------------------------------------|---------------------|----------------|--------|--|---|
| Recovery | | Spontaneous in le | ace than 2 coc | | A | | less than 2 coc | | A | | |
| • | | | ess than 3 sec | | | Spontaneous in less than 3 sec | | | | | |
| Dive forward angle on exit Change of course | | 0° - 30° Changing course | less than 45° | | A | Changing course | e less than 45° | | B A | | |
| Cascade occurs | | No No | | | A | No No | | | A | | |
| 12. High angle of attack recovery - 4.1.12 | | | | | | | | | | | |
| Recovery | | Spontaneous in I | Spontaneous in less than 3 sec | | Α | Spontaneous in | less than 3 sec | | Α | | |
| Cascade occurs | | No | | | A | No | | | Α | | |
| 13. Recovery from a developed full stall - 4.1.1 | 3 | | | | | | | | | | |
| Dive forward angle on exit | | 0° - 30° | | | A | 30° - 60° | | | В | | |
| Collapse Cascade occurs (other than collapse) | | No collapse No | | | A | No collapse No | | | A | | |
| Rocking backward | | Less than 45° | | | Α | Less than 45° | | | Α | | |
| Line tension | | Most lines tight | | | Α | Most lines tight | | | Α | | |
| 14. Asymmetric collapse (trim speed) - 4.1.14 | I | | | | | I | 1 | | | | |
| Change of course until re-inflation | apse | < 90° | Dive or roll angle | 0°- 15° | Α | < 90° | Dive or roll angle | 0°- 15° | А | | |
| Re-inflation behavior | beed colli | Spontaneous re-i | nflation | | Α | Spontaneous re- | -inflation | | Α | | |
| Total change of course | trim speed, max 50% collapse | Less than 360° | | | A | Less than 360° No | | | A | | |
| Collapse on the opposite side occurs Twist occurs | nax tr | No No | | | A | No No | | | A | | |
| Cascade occurs | - | No | | | A | No | | | A | | |
| Change of course until re-inflation | a) | 90°- 180° | Dive or roll angle | 15°- 45° | В | 90° - 180° | Dive or roll angle | 15° - 45° | В | | |
| Re-inflation behavior | trim speed, max 75% collapse | Spontaneous re-i | | - | A | Spontaneous re- | | | A | | |
| Total change of course | trim speed < 75% colla | Less than 360° | | | | Less than 360° | | | A | | |
| Collapse on the opposite side occurs | trim < 75° | No | | | A | No No | | | A | | |
| Twist occurs | may | No | | | Α | No | | | Α | | |
| Cascade occurs | | No | | | Α | No | | | Α | | |
| Change of course until re-inflation | Se | < 90° | Dive or roll angle | 15° - 45° | Α | < 90° | Dive or roll angle | 15° - 45° | А | | |
| Re-inflation behavior | accelerated, max 50% collapse | Spontaneous re-i | nflation | | Α | Spontaneous re- | -inflation | | А | | |
| Total change of course | seler 30% | Less than 360° | | | Α | Less than 360° | | | А | | |
| Collapse on the opposite side occurs | acc ax 5 | No | | | A | No | | | A | | |
| Twist occurs Cascade occurs | Ε . | No No | | | A A | No No | | | A | | |
| Change of course until re-inflation | ų. | 90°- 180° | Dive or roll angle | 15° - 45° | В | 90° - 180° | Dive or roll angle | 15° - 45° | В | | |
| Re-inflation behavior | accelerated, max 75% collapse | Spontaneous re-i | nflation | | А | Spontaneous re- | -inflation | <u> </u> | А | | |
| Total change of course | elera 5% c | Less than 360° | | | A | Less than 360° | | | Α | | |
| Collapse on the opposite side occurs | acc ax 7 | No | | | Α | No | | | Α | | |
| Twist occurs Cascade occurs | Ĕ | No No | | | A | No No | | | A | | |
| 15. Directional control with a maintained asymm | metric col | | | | | | | | | | |
| Able to keep course straight | | Yes | | | Α | Yes | | | А | | |
| 180° turn away from the collapsed side possible in | 10 sec | Yes | | | Α | Yes | | | Α | | |
| Amount of control range between turn and stall or | spin | More than 50% of the symmetric control travel | | | Α | More than 50% of | of the symmetric of | control travel | А | | |
| 16. Trim speed spin tendency - 4.1.16 | | | | | | l | | | | | |
| Spin occurs | | No | | | Α | No | | | Α | | |
| 17. Low speed spin tendency - 4.1.17 Spin occurs | | No | | | ^ | No | | | ^ | | |
| 18. Recovery from a developed spin - 4.1.18 | | 140 | | | Α | INO | | | А | | |
| Spin rotation angle after release | | Stops spinning in | less than 90° | | А | Stops spinning in | n less than 90° | | А | | |
| Cascade occurs | | No | | | A | No | | | A | | |
| 19. B-line-stall - 4.1.19 | | , | | | | , | | | | | |
| Change of course before release | | Changing course | less than 45° | | А | Changing course | e less than 45° | | А | | |
| Behaviour before release | | Remains stable w | vith straight span | | A Remains stable with straight span | | | А | | | |
| Recovery | | Spontaneous in le | ess than 3 sec | | A Spontaneous in less than 3 sec | | | А | | | |
| Dive forward angle on exit Cascade occurs | | 0° - 30° No | | | A A | | | | A A | | |
| 20. Big ears - 4.1.20 | | 140 | | | A | 140 | | | A | | |
| Entry procedure | | Special device re- | quired | | Α | Special device re | equired | | А | | |
| Behaviour during big ears | | Stable flight | | <u> </u> | | | A Stable flight | | | | Α |
| Recovery | | Recovery through pilot action in less than a further 3 sec | | | | | less than 3 sec | | Α | | |
| Dive forward angle on exit | | 0°-30° | | | Α | 0° bis 30° | | | А | | |
| 21. Big Ears in accelerated flight - 4.1.21 | | Chooled desides | quirod | | ^ | Special desire | auirod | | ^ | | |
| Entry procedure Behaviour during big ears | | Special device re- | quirea | | A | Special device re Stable flight | equired | | A | | |
| Recovery | | Recovery through | pilot action in le | ss than a further | than a further | | | | A | | |
| Dive forward angle on exit | | 3 sec 0°-30° | | | B Spontaneous in 3 to 5 sec | | | | A | | |
| Behaviour immediately after releasing the accelarate | ator while | | | | Α | | | | | | |
| maintaining big ears | | Stable flight | | | Α | Stable flight | | | А | | |
| 22. Behaviour exiting a steep spiral - 4.1.22 | | | | | | | | | | | |
| | | | | | | | | | | | |

Certification U-Turn BLACKLIGHT S/M

| Tendency to return to straight flight | Spontaneous exit | Α | Spontaneous exit | Α | | | | |
|---|--------------------------------------|----|--------------------------------------|----|--|--|--|--|
| Turn angle to recover normal flight | Less than 720°, spontaneous recovery | Α | Less than 720°, spontaneous recovery | А | | | | |
| 23. Alternative means of directional control - 4.1.23 | | | | | | | | |
| 180° turn achievable in 20 sec | Yes | Α | Yes | | | | | |
| Stall or spin occurs | No | Α | No | Α | | | | |
| 24. Any other flight procedure and/or configuration desc | ribed in the user's manual - 4.1.24 | | | | | | | |
| Procedure works as descibed | | NA | | NA | | | | |
| Procedure suitable for novice pilots | | NA | | NA | | | | |
| Cascade occurs | | NA | | NA | | | | |
| 25. Remarks of testpilot: | | | | | | | | |
| | | | | | | | | |
| | | | | | | | | |
| | | | | | | | | |
| | | | | | | | | |
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Flight Test-Report Stand - 08.04.2010 Seite 3

Certification U-Turn BLACKLIGHT M





EAPR e.V - Marktstr. 11 - D-87730 Bad Grönenbach - Germany

| | Minimum take off w | eight | Maximum take off w | veight |
|-------------------------|--------------------|-------|------------------------|--------|
| Date of testing | 09.03.12 | | 23.03.12 | |
| Testpilot | Mike Küng | | Tschofen Johannes | |
| Harness | Academy-Equipment | 120 | Academy Test Equipment | 1 |
| Pilot's take off weight | 80 kg | E | 110 kg | |

| Classification | В |
|----------------|---|
|----------------|---|



| Test-criteria | | 40977 | Evaluation | 40991 | Evaluation |
|--|-------------|----------------------------------|------------|----------------------------------|------------|
| 1. Inflation / take-off - 4.1.1 | | | | | |
| Rising behavior | | Smooth, easy and constant rising | Α | Smooth, easy and constant rising | А |
| Special take off technique required | | No | А | No | Α |
| 2. Landing - 4.1.2 | | | | | |
| Special landing technique required | | No | А | No | А |
| 3. Speeds in straight flight - 4.1.3 | | | | | |
| Trim speed more than 30km/h | | Yes | А | Yes | Α |
| Speed range using the controls larger than 10km/h | 1 | Yes | А | Yes | Α |
| Minimum speed | | Less than 25 km/h | Α | Less than 25 km/h | Α |
| 4. Control movement - 4.1.4 | | | | | |
| Max. weight in flight up to 80kg | | | - | | - |
| Max. weight in flight 80 to 100kg | | Increasing > 60cm | А | | - |
| Max. weight in flight greater than 100kg | | | - | Increasing >65 cm | Α |
| 5. Pitch stability exiting accelerated flight - 4.1. | 5 | | • | | <u> </u> |
| Dive forward angle on exit | | Dive forward less than 30° | А | Dive forward less than 30° | А |
| Collapse occurs | | No | А | No | Α |
| 6. Pitch stability operating controls during acce | elerated fl | ight - 4.1.6 | | | |
| Collapse occurs | | No | Α | No | Α |
| 7. Roll stability and damping - 4.1.7 | | | | | |
| Oscillations | | Reducing | А | Reducing | А |
| 8. Stability in gentle spirals - 4.1.8 | | | | | |
| Tendency to return to straight flight | | Spontaneous exit | А | Spontaneous exit | А |
| 9. Behaviour in a steeply banked turn - 4.1.9 | | | | | |
| Sink rate after two turns | | 12m/s to 14m/s | А | More than 14m/s | В |
| 10. Symmetric front collapse - 4.1.10 | | | <u> </u> | | <u> </u> |
| Entry | | Rocking back less than 45° | А | Rocking back less than 45° | А |
| Recovery | trim speed | Spontaneous in 3 to 5 sec | В | Spontaneous in less than 3 sec | А |
| Dive forward angle on exit | Ę. | 30°- 60° Keeping course | В | 0°-30° Keeping course | А |
| Cascade occurs | == | No | A | No | A |
| Entry | D | Rocking back less than 45° | А | Rocking back less than 45° | Α |
| Recovery | accelerated | Spontaneous in 3 to 5 sec | В | Spontaneous in less than 3 sec | Α |
| Dive forward angle on exit | CCe | 30° - 60° Keeping course | В | 30°- 60° Keeping course | В |
| Cascade occurs | Ø | No | А | No | Α |
| 11. Exiting deep stall (parachutal stall) - 4.1.11 | | | | | |

Certification U-Turn BLACKLIGHT M

| Deep stall achieved | | Yes | | | Yes | | | | | | |
|--|----------------------------------|---------------------------|---------------------------------------|----------------------------------|-------------------------------------|-----------------------------------|----------------------------------|---------------|---|--|---|
| Recovery | | Spontaneous in less | than 3 sec | | А | Spontaneous in | less than 3 sec | | А | | |
| Dive forward angle on exit | | 0°-30° | | | A | 30°- 60° | | | В | | |
| Change of course | | Changing course les | ss than 45° | | Α | | Changing course less than 45° No | | | | |
| Cascade occurs 12. High angle of attack recovery - 4.1.12 | | No | | | Α | INO | | | Α | | |
| Recovery | | Spontaneous in less | than 3 sec | | А | Spontaneous in | less than 3 sec | | А | | |
| Cascade occurs | | No Sportaneous in less | 0 000 | | A | No Sportaneous III | | | A | | |
| 13. Recovery from a developed full stall - 4.1.1 | 3 | 1 1 1 E | | | | L 11* | | | A | | |
| Dive forward angle on exit | | 0°-30° | | | Α | 30°-60° | | | В | | |
| Collapse Cascade occurs (other than collapse) | | No collapse No | · · · · · · · · · · · · · · · · · · · | | A | No collapse No | | | A | | |
| Rocking backward | | Less than 45° | | | Α | Less than 45° | | | Α | | |
| Line tension 14. Asymmetric collapse (trim speed) - 4.1.14 | | Most lines tight | | | А | Most lines tight | | | А | | |
| | | < 90° I | Dive or roll angle | 15° - 45° | А | < 90° | Dive or roll angle | 0°- 15° | А | | |
| Change of course until re-inflation | , apse | × 90 | Dive or roll angle | 15 - 45 | A | × 90 | Dive or roll angle | 0 - 15 | A | | |
| Re-inflation behavior | trim speed, max 50% collapse | Spontaneous re-infla | ation | | Α | Spontaneous re- | inflation | | Α | | |
| Total change of course Collapse on the opposite side occurs | nim s 50% | Less than 360° No | | | A | Less than 360° | | | A | | |
| Twist occurs | t max | No | | | A | No | | | A | | |
| Cascade occurs | | No | 1 | | Α | No | 1 | | Α | | |
| Change of course until re-inflation | se | 90° - 180° | Dive or roll angle | 15° - 45° | В | 90°- 180° | Dive or roll angle | 15° - 45° | В | | |
| Re-inflation behavior | trim speed, max 75% collapse | Spontaneous re-infla | ation | | А | Spontaneous re- | inflation | | А | | |
| Total change of course | n sp. | Less than 360° | | | Α | Less than 360° | | | А | | |
| Collapse on the opposite side occurs Twist occurs | trim lax 759 | No No | | | A | No No | | | A | | |
| Cascade occurs | Ε | No | | | A | No | | | A | | |
| Change of course until re-inflation | 99 | 90°- 180° | Dive or roll angle | 15° - 45° | В | < 90° | Dive or roll angle | 15° - 45° | А | | |
| Re-inflation behavior | accelerated, max 50% collapse | Spontaneous re-infla | ation | | А | Spontaneous re- | inflation | | А | | |
| Total change of course | celei 20% | Less than 360° | | | A | Less than 360° | | | Α | | |
| Collapse on the opposite side occurs Twist occurs | ac | No No | | | A | No No | | | A | | |
| Cascade occurs | _ | No | | | A | No | | | A | | |
| Change of course until re-inflation | I, | 90° - 180° | Dive or roll angle | 15° - 45° | В | 90°- 180° | Dive or roll angle | 15° - 45° | В | | |
| Re-inflation behavior | accelerated, max 75% collapse | Spontaneous re-infla | ation | | Α | Spontaneous re- | eous re-inflation | | Α | | |
| Total change of course | cele 75% | Less than 360° | | | A | Less than 360° No No | | | A | | |
| Collapse on the opposite side occurs Twist occurs | тах | No No | | | A | | | | A | | |
| Cascade occurs | 1 | No | | | А | No | | | А | | |
| 15. Directional control with a maintained asymmetry Able to keep course straight | metric col | Yes | | | A | Yes | | | A | | |
| 180° turn away from the collapsed side possible in | 10 sec | Yes | | | A | Yes | | | A | | |
| | | | | | | | | | | | |
| Amount of control range between turn and stall or 16. Trim speed spin tendency - 4.1.16 | spin | More than 50% of th | e symmetric co | ontrol travel | А | More than 50% of | of the symmetric o | ontrol travel | А | | |
| Spin occurs | | No | | | А | No | | | Α | | |
| 17. Low speed spin tendency - 4.1.17 | | | | | | | | | | | |
| Spin occurs 18. Recovery from a developed spin - 4.1.18 | | No | | | А | No | | | А | | |
| | | Stone eninnia - i- i | on then 00° | | | Stone enii- | n loop than 00° | | | | |
| Spin rotation angle after release | | Stops spinning in les | ss than 90° | | | A Stops spinning in less than 90° | | | A | | |
| Cascade occurs 19. B-line-stall - 4.1.19 | | No | | | Α | No | | | Α | | |
| Change of course before release | | Changing course les | ss than 45° | | Α | Changing course | e less than 45° | | Α | | |
| Behaviour before release | | Remains stable with | straight span | | A Remains stable with straight span | | | А | | | |
| Recovery | | Spontaneous in less | than 3 sec | | A Spontaneous in less than 3 sec | | | | А | | |
| Dive forward angle on exit | | 0°-30° | | | A | 0°-30° | | | A | | |
| Cascade occurs 20. Big ears - 4.1.20 | | No | | | А | No | | | Α | | |
| Entry procedure | | Special device requir | red | | А | Special device re | equired | | А | | |
| Behaviour during big ears | | Stable flight | | Stable flight | | | A Stable flight | | | | А |
| Recovery | | Spontaneous in 3 to 5 sec | | B Spontaneous in less than 3 sec | | | | Α | | | |
| Dive forward angle on exit | | 0°- 30° | | | Α | 0° bis 30° | | | А | | |
| 21. Big Ears in accelerated flight - 4.1.21 | | | | | | | | | | | |
| Entry procedure | | Special device required | | | Α | Special device re | equired | | А | | |
| Behaviour during big ears | | Stable flight | | | Α | Stable flight | | | Α | | |
| Recovery | | Spontaneous in 3 to | 5 sec | | Α | Spontaneous in | less than 3 sec | | А | | |
| Dive forward angle on exit | | 0°-30° | | | Α | 0° bis 30° | | | Α | | |
| Behaviour immediately after releasing the accelara maintaining big ears | ator while | Stable flight | | | Α | Stable flight | | | А | | |
| 22. Behaviour exiting a steep spiral - 4.1.22 | | | | | | | | | | | |
| | | | | | | | | | | | |

Certification U-Turn BLACKLIGHT M

| Tendency to return to straight flight | Spontaneous exit | А | Spontaneous exit | Α |
|--|--|--------------|--|-------------|
| Turn angle to recover normal flight | Less than 720°, spontaneous recovery A Less than 720°, spontaneous recovery | | Less than 720°, spontaneous recovery | Α |
| 23. Alternative means of directional control - 4.1.23 | | | | |
| 180° turn achievable in 20 sec | Yes | А | Yes | Α |
| Stall or spin occurs | No | А | No | Α |
| 24. Any other flight procedure and/or configuration desc | ribed in the user's manual - 4.1.24 | | | |
| Procedure works as descibed | | NA | | NA |
| Procedure suitable for novice pilots | | NA | | NA |
| Cascade occurs | | NA | | NA |
| 25. Remarks of testpilot: | | | | |
| | | | | |
| | | | | |
| | | | | |
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Certification U-Turn BLACKLIGHT L





EAPR e.V - Marktstr. 11 - D-87730 Bad Grönenbach - Germany

| | Minimum take off w | eight | Maximum take off weight | | |
|-------------------------|------------------------|-------|-------------------------|--|--|
| Date of testing | 17.05.12 | | 26.05.12 | | |
| Testpilot | Tschofen Johannes | | Anselm Rauh | | |
| Harness | Academy Test Equipment | 1 | EAPR Testequipment | | |
| Pilot's take off weight | 100 kg | | 125 kg | | |

| Classification | В |
|----------------|---|
|----------------|---|



| Test-criteria | | 41046 | Evaluation | 41056 | Evaluation |
|---|---|---|--------------------|----------------------------------|------------|
| 1. Inflation / take-off - 4.1.1 | | | | | |
| Rising behavior | | Smooth, easy and constant rising | А | Smooth, easy and constant rising | А |
| Special take off technique required | | No | А | No | А |
| 2. Landing - 4.1.2 | | | | | |
| Special landing technique required | | No | А | No | А |
| 3. Speeds in straight flight - 4.1.3 | | | | | |
| Trim speed more than 30km/h | | Yes | А | Yes | А |
| Speed range using the controls larger than 10km/ | h | Yes | А | Yes | А |
| Minimum speed | | Less than 25 km/h | Α | Less than 25 km/h | Α |
| 4. Control movement - 4.1.4 | | | | | |
| Max. weight in flight up to 80kg | | | - | | - |
| Max. weight in flight 80 to 100kg | | | - | | - |
| Max. weight in flight greater than 100kg | ax. weight in flight greater than 100kg | | А | Increasing >65 cm | А |
| 5. Pitch stability exiting accelerated flight - 4.1 | .5 | | | | |
| ive forward angle on exit | | Dive forward less than 30° A Dive forward less than 30° | | Dive forward less than 30° | А |
| ollapse occurs | | No A No | | No | А |
| 6. Pitch stability operating controls during acc | elerated f | light - 4.1.6 | | | |
| Collapse occurs | | No | Α | No | А |
| 7. Roll stability and damping - 4.1.7 | | | | | |
| Oscillations | llations | | Α | Reducing | Α |
| 8. Stability in gentle spirals - 4.1.8 | | | | | |
| Tendency to return to straight flight | | Spontaneous exit | A Spontaneous exit | | А |
| 9. Behaviour in a steeply banked turn - 4.1.9 | | | | | |
| Sink rate after two turns | | 12m/s to 14m/s | А | More than 14m/s | В |
| 10. Symmetric front collapse - 4.1.10 | | | | | |
| Entry | _ | Rocking back less than 45° | А | Rocking back less than 45° | А |
| Recovery | trim speed | Spontaneous in less than 3 sec | А | Spontaneous in less than 3 sec | А |
| Dive forward angle on exit | Ë | 0° - 30° Keeping course | А | 0° - 30° Keeping course | А |
| Cascade occurs | | No | Α | No | Α |
| Entry | D D | Rocking back less than 45° | Α | Rocking back less than 45° | A |
| Recovery | accelerated | Spontaneous in less than 3 sec | А | Spontaneous in less than 3 sec | Α |
| Dive forward angle on exit | 900 | 0° - 30° Keeping course | Α | 30° - 60° Keeping course | В |
| Cascade occurs | a | No | Α | No | Α |

Certification U-Turn BLACKLIGHT L

| r = | | | | | | | | | |
|--|----------------------------------|---|--------------------|-----------------------------|--|--------------------------------|----------------------|--------------------|--------|
| Deep stall achieved | | Yes | | | Yes | | | | |
| Recovery | | Spontaneous in less than 3 sec | | Α | | | | Α | |
| Dive forward angle on exit | | 0° - 30° Changing course less than 45° | | A | 30° - 60° Changing course less than 45° | | | В | |
| Change of course Cascade occurs | | No | | A | Changing course less than 45° No | | | A A | |
| 12. High angle of attack recovery - 4.1.12 | | • | | | | | | | |
| Recovery | | Spontaneous in I | ess than 3 sec | | А | Spontaneous in | less than 3 sec | | А |
| Cascade occurs | | No | | | A | No | | | A |
| 13. Recovery from a developed full stall - 4.1.1 | 3 | | | | | | | | - / . |
| Dive forward angle on exit | | 0° - 30° | | | А | 30° - 60° | | | В |
| Collapse Cascade occurs (other than collapse) | | No collapse No | | | A | No collapse No | | | A A |
| Rocking backward | | Less than 45° | | | A | Less than 45° | | | A |
| Line tension | | Most lines tight | | | А | Most lines tight | | | Α |
| 14. Asymmetric collapse (trim speed) - 4.1.14 | | | | | | | | | |
| Change of course until re-inflation | abse | < 90° | Dive or roll angle | 15° - 45° | А | < 90° | Dive or roll angle | 0° - 15° | Α |
| Re-inflation behavior | trim speed, max 50% collapse | Spontaneous re-i | inflation | | Α | Spontaneous re- | inflation | | Α |
| Total change of course | m sk 20% | Less than 360° | | | Α | Less than 360° | | | Α |
| Collapse on the opposite side occurs Twist occurs | tri ax (| No No | | | A | No No | | | A A |
| Cascade occurs | = | No | | | A | No | | | A |
| Change of course until re-inflation | | 90° - 180° | Dive or roll angle | 15° - 45° | В | 90° - 180° | Dive or roll angle | 15° - 45° | В |
| Re-inflation behavior | trim speed, max 75% collapse | Spontaneous re-i | | | A | Spontaneous re- | | - | A |
| Total change of course | trim speed, < 75% colla | Less than 360° | | | A | Less than 360° | | | A |
| Collapse on the opposite side occurs | trim x 75 | No No | | | A | No | | | A |
| Twist occurs | E B | No | | | Α | No | | | Α |
| Cascade occurs | | No | | | А | No | | | Α |
| Change of course until re-inflation | esc | < 90° | Dive or roll angle | 15° - 45° | А | < 90° | Dive or roll angle | 15° - 45° | Α |
| Re-inflation behavior | accelerated, max 50% collapse | Spontaneous re-i | inflation | | А | Spontaneous re-inflation | | | Α |
| Total change of course | cele 50% | Less than 360° | | | Α | Less than 360° | | | Α |
| Collapse on the opposite side occurs Twist occurs | ая | No No | | | A | No No | | | A A |
| Cascade occurs | = | No | | | A | No | | | A |
| Change of course until re-inflation | se | 90° - 180° | Dive or roll angle | 15° - 45° | В | 90° - 180° | Dive or roll angle | 15° - 45° | В |
| Re-inflation behavior | accelerated, max 75% collapse | Spontaneous re-i | inflation | | А | Spontaneous re-inflation | | | Α |
| Total change of course | cele 75% | Less than 360° | | | | Less than 360° | | | Α |
| Collapse on the opposite side occurs Twist occurs | ac nax . | No No | | | A | No No | | | A A |
| Cascade occurs | _ | No | | A | No | | | A | |
| 15. Directional control with a maintained asymmetry | metric col | lapse - 4.1.15 | | | | | | | |
| Able to keep course straight | | Yes | | | А | Yes | | | A |
| 180° turn away from the collapsed side possible in | 10 sec | Yes | | | Α | Yes | | | Α |
| Amount of control range between turn and stall or | spin | More than 50% of | f the symmetric c | ontrol travel | А | More than 50% of | of the symmetric c | ontrol travel | Α |
| 16. Trim speed spin tendency - 4.1.16 Spin occurs | | No | | | ۸ | No | | | ^ |
| 17. Low speed spin tendency - 4.1.17 | | 140 | | | Α | 140 | | | Α |
| Spin occurs | | No | | | А | No | | | A |
| 18. Recovery from a developed spin - 4.1.18 | | • | | | | • | | | |
| Spin rotation angle after release | | Stops spinning in | less than 90° | | А | Stops spinning i | n less than 90° | | А |
| Cascade occurs | | No | | | A | No | | | A |
| 19. B-line-stall - 4.1.19 | | | | | | | | | |
| Change of course before release | | Changing course | less than 45° | | А | Changing course | e less than 45° | | А |
| Behaviour before release | | Remains stable w | vith straight span | | А | Remains stable | with straight span | | Α |
| Recovery | | Spontaneous in I | ess than 3 sec | | Α | Spontaneous in less than 3 sec | | | Α |
| Dive forward angle on exit Cascade occurs | | 0° - 30° No | | | A | 30° - 60° No | | | A A |
| 20. Big ears - 4.1.20 | | | | | | | | | 7, |
| Entry procedure | | Special device re- | quired | | А | Special device re | equired | | Α |
| Behaviour during big ears | | Stable flight | | able flight A | | Stable flight | | А | |
| Recovery | | Spontaneous in 3 to 5 sec | | B Spontaneous in 3 to 5 sec | | | В | | |
| Dive forward angle on exit | d angle on exit 0° - 30° | | Α | 0° bis 30° | | | Α | | |
| 21. Big Ears in accelerated flight - 4.1.21 | | , | | | | | | | |
| Entry procedure | | Special device required | | A Special device required | | | Α | | |
| Behaviour during big ears | | Stable flight | | | Α | Stable flight | h nilot action in !- | ee than a firsthan | Α |
| Recovery | | Spontaneous in I | ess than 3 sec | | Α | 3 sec | h pilot action in le | ss than a further | В |
| Dive forward angle on exit | | 0° - 30° | 0° - 30° | | А | 0° bis 30° | | | Α |
| Behaviour immediately after releasing the accelara maintaining big ears | ator while | Stable flight | | | А | Stable flight | | | А |
| 22. Behaviour exiting a steep spiral - 4.1.22 | | | | | | | | | |
| | | | | | | | | | |

Certification U-Turn BLACKLIGHT L

| Tendency to return to straight flight | Spontaneous exit | А | Spontaneous exit | А |
|--|--|--------------|---|-------------|
| Turn angle to recover normal flight | Less than 720°, spontaneous recovery A Less than 720°, spontaneous recovery | | Less than 720°, spontaneous recovery | Α |
| 23. Alternative means of directional control - 4.1.23 | | | | |
| 180° turn achievable in 20 sec | Yes | Α | Yes | А |
| Stall or spin occurs | No | Α | No | Α |
| 24. Any other flight procedure and/or configuration desc | cribed in the user's manual - 4.1.24 | | | |
| Procedure works as descibed | | NA | | NA |
| Procedure suitable for novice pilots | | NA | | NA |
| Cascade occurs | | NA | | NA |
| 25. Remarks of testpilot: | | | | |
| | | | | |
| | | | | |
| | | | | |
| | | | | |
| | | | | |
| Copyright Ralf Antz 2010 | This Flight Te | est Report v | vas generated automatically and is valid withou | t signature |

Flight Test-Report Stand - 08.04.2010 Seite 3

<u>Instruction leaflet for repairs and 2 annual Check</u>

your airline...

U-Turn GmbH Im Neuneck 1 78609 Tuningen Germany

Tel: +49 (0)7464/9891280 Fax: +49 (0)7464/989128-28

Instruction leaflet for repairs and 2 annual Check

| Name: | |
|--|--------------------------------|
| Adress: | |
| Land: | Telephone Number: |
| E-Mail: | |
| Paraglider type and Color: | Serial number: |
| comments/notes: | |
| | |
| | |
| 2 annual Check | Line Check incl. strength test |
| Air permeability check | Repair of the marked damage |
| Recall with sighting of the paraglider | |
| O-TURN | CA-TURN |
| O C C | SSIII NATURINI |

Please, pretend the repair-destitute place in the top sail and / or bottom sail.

Line Order sheet



U-Turn GmbH Im Neuneck 1 78609 Tuningen Germany Tel: +49 (0)7464/9891280 Fax: +49 (0)7464/989128-28

LINE ORDER SHEET / BESTELLFORMULAR FÜR LEINEN

| Name | | | |
|-------------------------------------|-----------|-------------|-----------|
| Adress / Adresse | | | |
| | | | |
| E-mail | | inlin | |
| Telephone Number / Telefon Nummer | 101 | ir an | |
| Paragliding name / Gleitschirm Name | | 100 | |
| Size / Größe | | | |
| Other / Sonstiges | | | |
| | | inline | |
| | | ra | |
| | Jou | | |
| Serial Number / Serien | Nummer: _ | | -1 |
| Line ID / | Quantity/ | Line ID / | Quantity/ |
| Bezeichnung | Stückzahl | Bezeichnung | Stückzahl |
| | | | |
| | | Tine. | |
| | | airi | |
| | | 5 | |
| | 40 | | |
| | | | |

Business Replay Card

U-Turn GmbH Im Neuneck 1 D- 78609 Tuningen



| Name : |
|-------------------|
| First name: |
| Street: |
| Zip code/ City: |
| Telephone: |
| E-Mail: |
| Paraglider type: |
| Serial number: |
| Date of purchase: |
| DealershipP: |
| Tested by: |
| Flying hours: |
| Paraglider since: |
| Miscellaneous: |
| |

Yes, I would like to get the newsletter by E-Mail



Maintenance manual

as developer and manufacturer for paragliders, harnesses and rescue parachutes

Englisch Rev. 2.0 Stand: 12.2011

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Text and Graphics: Ernst Strobl

All technical details in this manual have been carefully checked by U-Turn. However we like to mention that we don't take any liability for possible mistakes, neither in legal responsibility, nor in liability cases that derive from mistakable details. We preserve the right to change this manual in any way to achieve technical improvements.

Topic of the inspection and reinspection intervals

Regular inspection according to aircraft inspection ordinance for standardized evaluated gliders. For school gliders afer 1 year, aircraft for recreational use after 2 years. Tandem gliders for commercial purposes annually, non commercial use every 2 years to be inspected. The inspection shall take place in the aforementioned intervals, or no later than 150 hours. Ground handling needs to be included in the sum of flight hours. Generally speaking: in the case any abnormal flight behavior, the manufacturer should be informed and the canopy, if necessary, sent in for inspection.

Who may inspect/test?

Besides the manufacturer or the by him approved person or instance is authorized the owner of the glider to warrant the bi-annual inspection and only if in compliance with pre-requisites set forth.

Individual personal prerequisites for the inspections

Personal prerequisites for the inspection of individually owned solo gliders for recreational use only:

- Holder of a valid unrestricted license for paragliders or equivalent accredited license.
- An adequate orientation in the operation by the manufacturer. For this a 3 month formation with
- the manufacturer is necessary.
- If a glider was tested for personal use exclusively, then its use by a third party is not allowed.

Individual personal prerequisites for the inspection of gliders, RG,GZ, used by third parties or for tandem purpose:

- A for the testing prescribed professional training.
- A vocational activity in the productin or maintenance of GS, RG, GZ or one of a technically similiar nature. Of which 6 month within the last 24 in a manufacturing operation recreational free
- flight aircraft.
- An at least 2 week, subject to charge, relevant training course at the operation of the manufacturer.
- An applicable orientation for each type of device, which is to be refreshed annually.

Necessary equipment and documentation

- Gauge, preferably Kretschmer (brand) with manual.
- Bettsometer with manual.
- Maintenance directions by manufacturer.
- Original materials and -spare parts, as well as original material-record for the device.
- Assertion of airworthiness for the device.
- Airsports device identification tag (see manual).
- Line length table (see manual).
- Line lenght logs (if available).
- Inspection log (collecting main) to the documentation.
- Lighttable for visual inspection of the reserve.

During the inspection the following steps are to be taken in:

Positive identification of the device:

Positive identification of the aircraft (Type, size, etc.) on the basis certification seal or placard..

- Are the pertinent manufacturer documents available?
- If certification seal and/ or placard are in place, are they readable and correct?
- If not so: Please obtain from manufacturer or dealer in question.

The determined values / modification are to be noted in an inspection log!

Inspection of the reserve parachute

Before packing the reserve parachute this is to be checked by packer. If the parachute was deployed for a rescue, then it is subject to an inspection. If a folded reserve parachute is re-packed again a deployment check is to be staged, to be determined is if the force for deployment is between a minimum of 3kg and maximum of 6kg.

Testing of the topsail, undersail, seams, reserve parachute of

holes and tears

The topsail and undersail of both paragliders as well as reserve parachutes must, for each cell (paragliders) and each gore (parachutes), from the leading edge to the trailing edge, submitted to the following checks. If in one of the following attributes anomalies are discovered, the glider is to be sent in to the manufacturer for inspection.

- Check for holes smaller or larger tears, deformations and abraded areas.
- Defeciencies in the coating, other aberrations in the canopy like e.g. old repairs.
- With reserve parachutes a light-talbe is to be used for an inspection for holes, tears and deformations.

Abrasison and deformaties

With large and critical abrasion and deformations, the entire cell panel in question must be replaced by the manufacturer. The determined values/modifications are to be noted in the testin log!

Testing of the ribs

Visual inspection of the chambers (from the leading to the trailing edge) whether the stitching in the seams, cell partition ribs and reinforcements are in good shape, thus without tears, deformations, abrasions or damage of the coating.

With torn ribs, defective, loose or missing stitching in the seams the glider must be returned the to the manufacturer or authorized inspection operation. The determined values/modifications are to be noted in the inpection log!

Check of the tear resistance

To be conducted with the Bettsometer at the following points (B.M.A.A. approved patent number GB2270768 Clive of bed Sails)

The test sequence is to be inferred from the operating instruction the Bettosometer.

- In both the top and undersail where the A-lines connect, push a needle-thick hole and check
- the tear resistance.
- The limit value of the measurement is determined at 500g, and a tear width of fewer than 5mm. The determined values / modifications are to be noted in the inspection log!

Porositycheck of the canopy

At all following measuring points the air porosity has to be more than at least 20 sec. (by Kretschmer).

At smaller air permeability values the paraglider must be returned to the manufacturer. Measuring points: The porosity measurements by the Kretschmer measuring method (please consider operating instruction) are to be conducted at the following points on the canopy check on both under and upper sail.

- Center cell approx. 20-30cm back from leading edge
- 3rd Cell off center both to the left /right approx. 20-30cm back from leading edge
- 10th Cell off center both to the left /right approx. 20-30 cam back from leading edge The determined values / modifications are to be noted in the inspection log!

Connection parts

Check of the webbing and maillons

- are there abrasions, buckling, tears, strong signs of wear obvious?
- Is all the stitching fast and firm?
- Is the accelerator running free and intact?
- Are brake toggle attachments still firmly sewn on?
- Are the maillons corrosion free, are the sleeves of the gates free moving on the thread? Measure under a load of 5 kg. The determined values are to be compared with the specifications from the EAPR-Technical data sheet. Allowable variations are to be inferred from the manufactu-

rer directions. If the webbing or parts thereof are defective, spare parts are to be ordered from the

manufacturer and replace the defective parts with original parts. The determined values/modification are to be note in the inspection log!

Lines

Test of the line tensile strength:

Line selection: select a middle, lower cascade of the A, B and a C- lines as well as if available a middle A and B upper cascade, and stress test for tensile strength testing device on their tensile strength. Tension velocity of the tension cylinder: v=30cm/min

Tear/tensile strength values:

the determined values/modifications are to be noted in the inspection!

Attention: Each size (line diameter) is to be assigned a fixed value.

In case the lines cannot withstand the indicated load/stress or pass tensile strength test, all other lines must also be changed. If the checked lines fulfill the test criteria, only those are replaced by new lines. All replaced lines are to be marked in the proximity of the maillon (seam) with a black felt marker pen and noted in the inspection log with the date of the exchange and the logged of hours of flight time of the glider. During the next test for tensile strength an original line, neighbouring the replaced line is to be sampled. The various line diameters are allocated a minimal Sewing lenght!

Check of the line length and line attachments

Bottom cascade, upper cascades and brake lines for, breaks, abrasions, visual check. First the A-lines, then B. etc.

- Are all lines adequately sewn and attached to the line attachments?
- Is the sheathing of the lines even are exactly?
- Are all loops, knots, seams in good shape?
- Are there any abrasions present?

Measuring the line lengths:

- The lines must be measured with a load of 5 kg, in order to obtain comparable results. The relevant line lengths are in the technical data sheet of the uster manual.
- The measurement takes place in accordance with DHV method, from the maillon to the canopy (inclusive attachment loop at the sail).
- The numbering takes place from the stabilo toward the center. Measuring the opposite facing of the wing can under same conditions also be conducted by a symmetry comparision.
- The results are again noted the inspection log and should be compared side by side to line lenghts of the EAPR technical data sheet. The tolerance in deviation of these values should not exceed more than +/-1,5cm
- If a line is defective, it is to be exchanged immediately. Please acquire the identification reference marking of the line from the line plan, order from the manufacturer and replace accordinglyor have it replaced.

The determined values / modifications are to be noted in the inspection log!

Occassional check of trim and adjustment

Before a test flight a visual inspection of the canopy and lines is to be conducted with the glider laid out as well as pulled up inflated.

In particular attention should be paid to the length of the brake lines with the canopy inflated. Only if all doubts are cleared concerning faulty adjustment of the brake lines, a check flight may be conducted.

Description of the materials and technical data

See manual of your paraglider.

miscellaneous

- All measurement and repair work at paraglider and rescue system must be documented completely in the inspection log.
- When packing or repacking the reserve parachute, special attention is to be paid to the particular packing directions of the manufacturer! See rescue / reserve equipment manual.
- With the exchange of parts or component modules only original materials or original replacement parts may be used!
- With sewing work the original sewing pattern is to be kept, patching and thread material of same
- strength and quality as original!
- The inspection survey and/or test log must with be signed, complete with place and date!
- The period for recordkeeping is 4 years.

Completed check very important!

Before you perform any checks and/or repairs yourself on your glider, we ask to read you the following pages carefully. You inform yourself hereby about prerequisites and conditions of a done in person bi-annual inspection.

- According to new DHV regulation, the customer (Glider-owner) can conduct the 2-yearly check-of the canopy with the help of the inspection directions and all necessary testing equipment and-documents in person on his own responsibility. In addition the wing does not have to be sent in to the manufacturer.
- The 2-yearly check may only be conducted by the glider owner personally, if he fulfills the prerequisites, or an inspection station authorized by the manufacturer. Inquire therefore with the manufacturer on authorized inspection stations.
- The owner of the canopy must be aware of the responsibility, which he takes with a self conducted 2-yearly check of the glider. The self performed 2-yearly check is only legally effective, if this is acknowledged after the check with date, name (in capitals) and signature on or beside the placard.
- Reserve equipment re-packing interval in accordance with DHV: Every 4 months a repacking is required. Allowed period of operation: 8 years, afterwards up to 12 years with an annual check
- About insurance-legal consequences of your self performed 2-yearly inspection you should inform with your insurer in a timely fashion.
- An inspection is valid only if the inspection log is completely filled out. Inform also about possible revisions of the inspection directions with the manufacturer before the inspection.
- Important: If the necessary efforts for the maintenance inspection cannot be carried out (required equipment and documents), should the canopy be sent in to the manufacturer.
- For paragliders, harnesses and reserve parachutes, which are checked, controlled, repaired, packed or repacked, test-flown and/or other maintenance work, by none U-turn authorized personnel forgot any guarantee and or warranty!
- All maintenance work must in be accordance with the maintenance specifications of the operation manual and the special maintenance directions of the manufacturer and the publications of the IHB to be conducted.
- With any abnormal appearances during the performance of maintenance is the technical manager to be informed, who has to decide on how to procede.
- With the replacement of parts or component modules only original materials or original partymay be used!